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Another gravity gradient satellite like Transit, but not in 60's in 80's
 Polar BEAR: Beacon Experiment + Auroral Research

rather than permanent magnets \rightarrow momentum wheel for pitch axis
 3 months after launch Solar Radiation flipped the satellite over
 and it re-stabilized upside down

to fix they stopped momentum wheel, de-stabilized, re-spin the wheel and it flipped back over to the correct position

$$H = J\omega$$

3×1 \uparrow B frame \uparrow B w/ respect to inertial frame
 B frame

ex: $\begin{bmatrix} 1 \\ 2 \\ 3 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 3 \end{bmatrix} \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix}$

$H + \omega$ do not point in the same direction unless $\lambda\omega = J\omega$

principal moments of Inertia (Eigenvalues): $\det(\lambda I - J) = 0$

Principal axes (Eigen vectors): $(\lambda_i I - J)e_i = 0$

see example

Now we have the B frame and E frame (principal axes)

$$R^{E/B} = \begin{bmatrix} e_{11} & e_{12} & e_{13} \\ e_{21} & e_{22} & e_{23} \\ e_{31} & e_{32} & e_{33} \end{bmatrix} = \begin{bmatrix} -1/\sqrt{2} & 1/\sqrt{2} & 0 \\ 0 & 0 & 1 \\ 1/\sqrt{2} & 1/\sqrt{2} & 0 \end{bmatrix}$$

$$H_E = R^{E/B} H_B \quad \omega_E = R^{E/B} \omega_B \rightarrow \omega_B = (R^{E/B})^T \omega_E$$

$$H_E = R^{E/B} H_B = R^{E/B} J_B \omega_B = R^{E/B} J_B (R^{E/B})^T \omega_E$$

$$\begin{bmatrix} -1/\sqrt{2} & 1/\sqrt{2} & 0 \\ 0 & 0 & 1 \\ 1/\sqrt{2} & 1/\sqrt{2} & 0 \end{bmatrix} \begin{bmatrix} 2 & 1 & 0 \\ 1 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix} \begin{bmatrix} 1/\sqrt{2} & 0 & 1/\sqrt{2} \\ 1/\sqrt{2} & 0 & 1/\sqrt{2} \\ 0 & 1 & 0 \end{bmatrix}$$

$$J_E = R^{E/B} J_B (R^{E/B})^T \text{ diagonalization} = \begin{bmatrix} -1/\sqrt{2} & 1/\sqrt{2} & 0 \\ 0 & 0 & 2 \\ 1/\sqrt{2} & 1/\sqrt{2} & 0 \end{bmatrix} \begin{bmatrix} 1/\sqrt{2} & 0 & 1/\sqrt{2} \\ 1/\sqrt{2} & 0 & 1/\sqrt{2} \\ 0 & 1 & 0 \end{bmatrix}$$

$$H_E = J_E \omega_E$$

Transform 1 moment of inertia matrix to another using Rotation Matrix

$$= \begin{bmatrix} 1 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 3 \end{bmatrix}$$

$$\vec{F} = \frac{d\vec{L}}{dt}$$

$$\vec{M} = \frac{d\vec{H}}{dt} = \frac{d\vec{H}}{dt} + \omega \times \vec{H}$$

All in frame B

$$M = \dot{H} + S(\omega)H$$

$$\begin{bmatrix} M_1 \\ M_2 \\ M_3 \end{bmatrix} = \begin{bmatrix} \dot{H}_1 \\ \dot{H}_2 \\ \dot{H}_3 \end{bmatrix} + \begin{bmatrix} 0 & -\omega_3 & \omega_2 \\ \omega_3 & 0 & -\omega_1 \\ -\omega_3 & \omega_1 & 0 \end{bmatrix} \begin{bmatrix} H_1 \\ H_2 \\ H_3 \end{bmatrix}$$

$$M = \dot{H} + S(\omega)J\omega$$

$$H = J\dot{\omega} + J\omega$$

rigid body

$$M = J\dot{\omega} + S(\omega)J\omega$$

J is positive definite \rightarrow Invertible

$$\dot{\omega} = J^{-1}(M - S(\omega)J\omega)$$

$$\dot{Q} = \frac{1}{2} \mathcal{R} \otimes Q$$

Finding principal axes

For a moment of inertia matrix J written with respect to frame B , each root $\lambda_1, \lambda_2, \lambda_3$ of

$$\det(\lambda I - J) = 0$$

is a principal moment of inertia, and each vector e_i satisfying

$$(\lambda_i I - J)e_i = 0$$

defines the coordinates of a principal axis with respect to frame B . For example, consider the inertia matrix

$$J = \begin{bmatrix} 2 & 1 & 0 \\ 1 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix}$$

1. Find the principal moments of inertia. (Work this out by hand.)

~~$2(\lambda-2) - 1(\lambda-2) = 0$~~

$$\begin{bmatrix} \lambda-2 & 1 & 0 \\ 1 & \lambda-2 & 0 \\ 0 & 0 & \lambda-2 \end{bmatrix}$$

~~$(\lambda-2)(\lambda-2)(\lambda-2) = 0 \Rightarrow (\lambda-2)^3 = 0$~~

~~$(\lambda^2 - 4\lambda + 4)(\lambda-2) - \lambda + 2$~~

~~$\lambda^3 - 6\lambda^2 + 12\lambda - 8 - \lambda + 2$~~

~~$\lambda^3 - 6\lambda^2 + 11\lambda - 6 = 0$~~

$$0 - 0 + \lambda - 2(\lambda-2)(\lambda-2) = 0$$

$$(\lambda-2)(\lambda^2 - 4\lambda + 3) = 0$$

$$(\lambda-2)(\lambda-3)(\lambda-1) = 0$$

$\lambda_1 = 2$ $\lambda_1 = 1$
 $\lambda_2 = 3$ $\lambda_2 = 2$
 $\lambda_3 = 1$ $\lambda_3 = 3$

2. Find the principal axis corresponding to the minor moment of inertia. (Again, do this by hand.)

~~$\begin{bmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix}$~~

~~$\begin{bmatrix} 2 & 1 & 0 \\ 1 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix}$~~

~~$\begin{bmatrix} 0 & 1 & 0 \\ 1 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$~~

~~$\begin{bmatrix} \lambda \\ \lambda \\ \lambda \end{bmatrix}$~~

~~$\lambda_1 = \lambda_2 = 0$ $\lambda_3 = 2$ $e_1 = \frac{1}{\sqrt{2}} \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix}$~~

or make $e_1 = \begin{bmatrix} -1/\sqrt{2} \\ 1/\sqrt{2} \\ 0 \end{bmatrix}$

so make $e_3 = \begin{bmatrix} -1/\sqrt{2} \\ -1/\sqrt{2} \\ 0 \end{bmatrix}$

$e_1 = \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix}$ $J e_1 = \begin{bmatrix} 2 & 1 & 0 \\ 1 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix} \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix} = \begin{bmatrix} 3 \\ 3 \\ 0 \end{bmatrix}$ $\lambda_1 e_1 = 1 \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix} = \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix} \neq \begin{bmatrix} 3 \\ 3 \\ 0 \end{bmatrix}$

$(\lambda_1 I - J)e_1 = 0$

$\begin{pmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{pmatrix} - \begin{pmatrix} 2 & 1 & 0 \\ 1 & 2 & 0 \\ 0 & 0 & 2 \end{pmatrix} \begin{bmatrix} e_{11} \\ e_{12} \\ e_{13} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$

$\begin{bmatrix} -1 & -1 & 0 \\ -1 & -1 & 0 \\ 0 & 0 & -1 \end{bmatrix} \begin{bmatrix} e_{11} \\ e_{12} \\ e_{13} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$

$e_{13} = 0$ $e_1 = e_2 \rightarrow \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix}$

$\begin{bmatrix} -1 \\ 1 \\ 0 \end{bmatrix}$ ✓ normalize: $\begin{bmatrix} 1/\sqrt{2} \\ -1/\sqrt{2} \\ 0 \end{bmatrix}$

$e_2 = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$

$e_3 = \begin{bmatrix} 1/\sqrt{2} \\ 1/\sqrt{2} \\ 0 \end{bmatrix}$

$\omega \times r = S(\omega) r$

$S(e_1) e_2 = \begin{bmatrix} 0 & \omega_1 e_{22} \\ \omega_2 e_{21} & 0 \\ \omega_3 e_{23} & \omega_4 e_{24} \end{bmatrix} \begin{bmatrix} e_{21} \\ e_{22} \\ e_{23} \end{bmatrix}$

$= \begin{bmatrix} 0 & 0 & -\omega_2 \\ 0 & 0 & \omega_1 \\ \omega_3 & -\omega_4 & 0 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} = \begin{bmatrix} -\omega_2 \\ \omega_1 \\ 0 \end{bmatrix} = e_3$